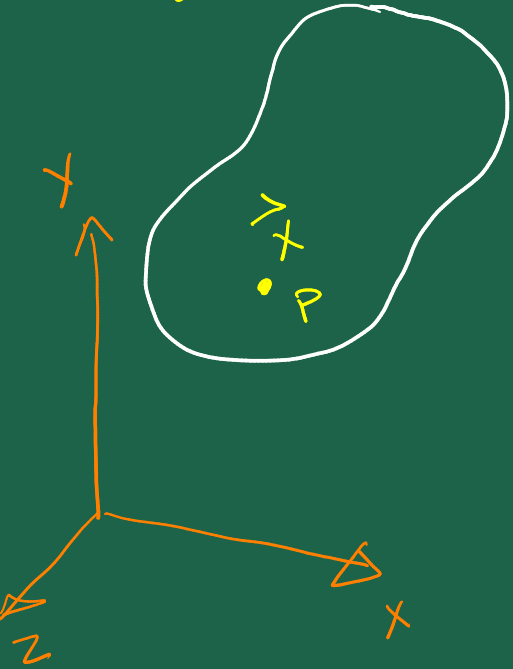


# Strain and Strain Transformation

$\Delta$   $\nabla$  nabla

Deformation



$$\vec{x}' \equiv \vec{x}(\vec{x})$$

Deformation map

$$\nabla \phi$$

$$\nabla \phi \equiv \frac{\partial \phi}{\partial x} \hat{i} + \frac{\partial \phi}{\partial y} \hat{j} + \frac{\partial \phi}{\partial z} \hat{k}$$

$$[\nabla \phi] \equiv \begin{bmatrix} \frac{\partial \phi}{\partial x} \\ \frac{\partial \phi}{\partial y} \\ \frac{\partial \phi}{\partial z} \end{bmatrix}$$

$$[\vec{v}] \equiv \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix}$$

$$[\nabla \vec{v}]$$

$$[\nabla \vec{v}] = \begin{bmatrix} \frac{\partial v_x}{\partial x} & \frac{\partial v_x}{\partial y} & \frac{\partial v_x}{\partial z} \\ \frac{\partial v_y}{\partial x} & \frac{\partial v_y}{\partial y} & \frac{\partial v_y}{\partial z} \\ \frac{\partial v_z}{\partial x} & \frac{\partial v_z}{\partial y} & \frac{\partial v_z}{\partial z} \end{bmatrix}$$

$$\vec{a} \cdot \vec{b} = \begin{bmatrix} a_x \\ a_y \\ a_z \end{bmatrix}^T \begin{bmatrix} b_x \\ b_y \\ b_z \end{bmatrix} = a_x b_x + a_y b_y + a_z b_z$$

$\vec{a} \cdot \nabla \vec{v} \rightarrow$  vector entity

$$([\vec{a}]^T [\nabla \vec{v}])^T \rightarrow 1 \times 3 \text{ (row matrix)} \quad \times$$

$$[\nabla \vec{v}] [\vec{a}] \rightarrow 3 \times 1 \quad \checkmark$$

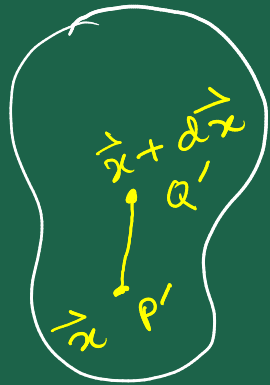
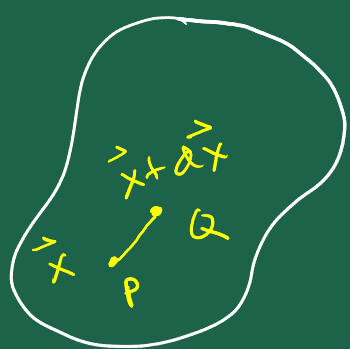
$$[\nabla \vec{v}]^T [\vec{a}] \rightarrow 3 \times 1 \quad \times$$

$$d\vec{x} \cdot \nabla \vec{u} = [\nabla \vec{u}] [d\vec{x}]$$

## Displacement

$$\vec{u} := \vec{x} - \vec{x}$$

## Quantification of deformation



$$\begin{aligned}\vec{u}(\vec{x} + d\vec{x}) &= (\vec{x} + d\vec{x}) - (\vec{x} + d\vec{x}) \\ &= \vec{x} - \vec{x} + d\vec{x} - d\vec{x} \\ &= \vec{u}(\vec{x}) + d\vec{x} - d\vec{x}\end{aligned}$$

$$\therefore d\vec{x} = d\vec{x} + \vec{u}(\vec{x} + d\vec{x}) - \vec{u}(\vec{x})$$

Taylor expansion

$$f(x+h) = f(x) + \frac{h}{L} f'(x) + \frac{h^2}{L^2} f''(x) + \dots$$

$$\begin{aligned}f(x+h, y+k) &= f(x, y) + \frac{h}{L} \frac{\partial f}{\partial x} + \frac{k}{L} \frac{\partial f}{\partial y} \\ &+ \frac{h^2}{L^2} \frac{\partial^2 f}{\partial x^2} + \frac{k^2}{L^2} \frac{\partial^2 f}{\partial y^2} + 2 \frac{hk}{L^2} \frac{\partial^2 f}{\partial x \partial y} \\ &+ \dots\end{aligned}$$

$$f(x+h, y+k, z+m) = f(x, y, z) + \frac{h}{L} \frac{\partial f}{\partial x} + \frac{k}{L} \frac{\partial f}{\partial y} + \frac{m}{L} \frac{\partial f}{\partial z} + \dots$$

$$f(\vec{x} + d\vec{a})$$

$$\vec{u}(\vec{x} + d\vec{x}) = \vec{u}(x+dx, y+dy, z+dz) \quad \left| \begin{array}{l} \vec{x} = x\hat{i} + y\hat{j} + z\hat{k} \end{array} \right.$$

$$= \vec{u}(x, y, z) + \frac{dx}{1} \frac{\partial \vec{u}}{\partial x} + \frac{dy}{1} \frac{\partial \vec{u}}{\partial y} + \frac{dz}{1} \frac{\partial \vec{u}}{\partial z} + \dots$$

$$= \vec{u}(\vec{x}) + dx \frac{\partial \vec{u}}{\partial x} + dy \frac{\partial \vec{u}}{\partial y} + dz \frac{\partial \vec{u}}{\partial z}$$

$$= \vec{u}(\vec{x}) + \left\{ (dx\hat{i} + dy\hat{j} + dz\hat{k}) \cdot \left( \frac{\partial}{\partial x}\hat{i} + \frac{\partial}{\partial y}\hat{j} + \frac{\partial}{\partial z}\hat{k} \right) \right\} \vec{u}$$

$$= \vec{u}(\vec{x}) + d\vec{x} \cdot \nabla \vec{u}$$

$$\therefore d\vec{x} = d\vec{x} + \cancel{\vec{u}(\vec{x} + d\vec{x})} - \cancel{\vec{u}(\vec{x})} = d\vec{x} + \vec{u}(\vec{x}) + d\vec{x} \cdot \nabla \vec{u} - \vec{u}(\vec{x})$$

$$= d\vec{x} + d\vec{x} \cdot \nabla \vec{u}$$

$$[d\vec{x}] = [d\vec{x}] + [d\vec{x} \cdot \nabla \vec{u}]$$

$$= [d\vec{x}] + [\nabla \vec{u}] [d\vec{x}] = [d\vec{x}] ([\mathbf{I}] + [\nabla \vec{u}])$$

$$|d\vec{x}| = \sqrt{dx^{\tilde{x}} + dy^{\tilde{y}} + dz^{\tilde{z}}}$$

$$|d\vec{x}| = \sqrt{dx^{\tilde{x}} + dy^{\tilde{y}} + dz^{\tilde{z}}}$$

$$|d\vec{x}| \leftrightarrow |d\vec{x}|$$

$$|d\vec{x}|^{\tilde{r}} \text{ with } |d\vec{x}|^{\tilde{r}}$$

$$|d\vec{x}|^2 = d\vec{x} \cdot d\vec{x}; |d\vec{x}|^{\tilde{r}} = d\vec{x} \cdot d\vec{x}$$

$$\sigma_{nn} \text{ or } T_{nn}$$

$$\sigma_{ns} \text{ or } T_{ns}$$

$$\varepsilon := \frac{|d\vec{x}| - |d\vec{x}|}{|d\vec{x}|}$$

$$|d\vec{x}|^2 = d\vec{x} \cdot d\vec{x}$$

$$= (d\vec{x} + d\vec{x} \cdot \nabla \vec{u}) \cdot (d\vec{x} + d\vec{x} \cdot \nabla \vec{u})$$

$$= [d\vec{x} + d\vec{x} \cdot \nabla \vec{u}]^T [d\vec{x} + d\vec{x} \cdot \nabla \vec{u}]$$

$$= ([d\vec{x}] + [\nabla \vec{u}][d\vec{x}])^T ([d\vec{x}] + [\nabla \vec{u}][d\vec{x}])$$

$$= ([d\vec{x}]^T + [d\vec{x}]^T [\nabla \vec{u}]^T) ([d\vec{x}] + [\nabla \vec{u}][d\vec{x}])$$

$$= [d\vec{x}]^T [d\vec{x}] + [d\vec{x}]^T [\nabla \vec{u}][d\vec{x}] + [d\vec{x}]^T [\nabla \vec{u}]^T [d\vec{x}]$$

$$+ \underbrace{[d\vec{x}]^T [\nabla \vec{u}]^T [\nabla \vec{u}][d\vec{x}]}_{\text{neglect}}$$

$$\approx |d\vec{x}|^2 + [d\vec{x}]^T ([\nabla \vec{u}] + [\nabla \vec{u}]^T) [d\vec{x}]$$

$$\vec{a} \cdot \vec{b}$$

$$[\vec{a}]^T [\vec{b}]$$

$$[d\vec{x} \cdot \nabla \vec{u}]$$

$$= [\nabla \vec{u}][d\vec{x}]$$

$$([A][B])^T$$

$$= [B]^T [A]^T$$

$$\left(\frac{\partial u}{\partial x}\right)^2, \frac{\partial u}{\partial x} \frac{\partial v}{\partial x}$$

$$\frac{\partial v}{\partial y} \frac{\partial w}{\partial z}$$

$$|d\vec{x}'|^2 = |d\vec{x}|^2 + [d\vec{x}]^T \left( [\nabla \vec{u}] + [\nabla \vec{u}]^T \right) [d\vec{x}]$$

Define the infinitesimal or small strain tensor

$$\underline{\underline{\varepsilon}} := \frac{1}{2} \left\{ \nabla \vec{u} + (\nabla \vec{u})^T \right\}$$

$$[\underline{\underline{\varepsilon}}] := \frac{1}{2} \left\{ [\nabla \vec{u}] + [\nabla \vec{u}]^T \right\}$$

↳ Symmetric part of  $[\nabla \vec{u}]$

$$[\underline{\underline{\varepsilon}}] = \begin{bmatrix} \varepsilon_{xx} & \varepsilon_{xy} & \varepsilon_{xz} \\ \varepsilon_{xy} & \varepsilon_{yy} & \varepsilon_{yz} \\ \varepsilon_{xz} & \varepsilon_{yz} & \varepsilon_{zz} \end{bmatrix}$$

= In terms of derivatives of  $u, v, w$

$$\varepsilon_{xx} = \frac{1}{2} \left( \frac{\partial u}{\partial x} + \frac{\partial u}{\partial x} \right) = \frac{\partial u}{\partial x}$$

$$\varepsilon_{xy} = \frac{1}{2} \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right), \dots$$

$$\begin{aligned} & \frac{1}{2} \left( [\nabla \vec{v}] + [\nabla \vec{v}]^T \right) \\ &= \underline{\underline{E}} \text{ (strain rate tensor)} \end{aligned}$$

$$\begin{aligned} & [A] \\ &= \frac{1}{2} \left( [A] + [A]^T \right) \\ &+ \frac{1}{2} \left( [A] - [A]^T \right) \\ &\rightarrow \text{Symmetric part} \end{aligned}$$

$$\begin{aligned} & \left\{ \frac{1}{2} \left( [A] + [A]^T \right) \right\}^T \\ &= \frac{1}{2} \left( [A]^T + [A] \right) \end{aligned}$$

$$|d\vec{x}'|^2 = |d\vec{x}|^2 + [d\vec{x}]^T \left( [\nabla \vec{u}] + [\nabla \vec{u}]^T \right) [d\vec{x}]$$

$$\Rightarrow |d\vec{x}'|^2 = |d\vec{x}|^2 + 2[d\vec{x}]^T [\underline{\underline{\varepsilon}}] [d\vec{x}]$$

$$\Rightarrow |d\vec{x}'|^2 = |d\vec{x}|^2 + 2|d\vec{x}|^2 [\hat{N}]^T [\underline{\underline{\varepsilon}}] [\hat{N}]$$

$$\Rightarrow \frac{|d\vec{x}'|^2}{|d\vec{x}|^2} = 1 + 2[\hat{N}]^T [\underline{\underline{\varepsilon}}] [\hat{N}]$$

$$\Rightarrow (1 + \varepsilon)^2 = 1 + 2[\hat{N}]^T [\underline{\underline{\varepsilon}}] [\hat{N}]$$

$$\Rightarrow 1 + 2\varepsilon + \varepsilon^2 = 1 + 2[\hat{N}]^T [\underline{\underline{\varepsilon}}] [\hat{N}]$$

Neglect  $\varepsilon^2$  compared  $\varepsilon$

$$\therefore \boxed{\varepsilon_N = [\hat{N}]^T [\underline{\underline{\varepsilon}}] [\hat{N}]}$$

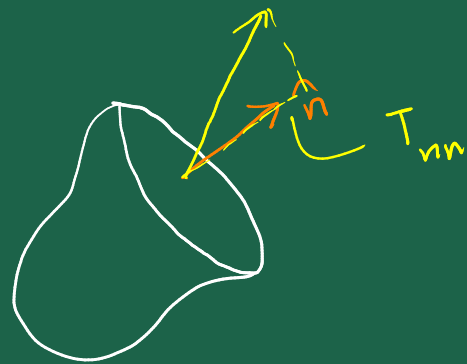
(Normal strain along  
a particular dir<sup>n</sup>  $\rightarrow \hat{N}$ )

$$d\vec{x} = |d\vec{x}| \hat{N}$$

$$[d\vec{x}] = |d\vec{x}| [\hat{N}]$$

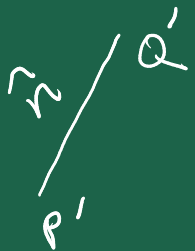
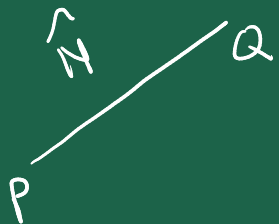
$$[d\vec{x}]^T = |d\vec{x}| [\hat{N}]^T$$

$$\varepsilon := \frac{|d\vec{x}'| - |d\vec{x}|}{|d\vec{x}|}$$

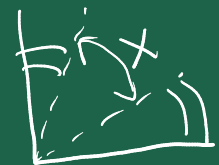
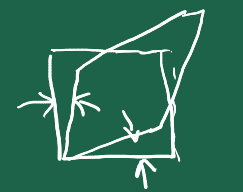




# Change in dir<sup>n</sup> of an elemental line segment



Given:  $\hat{N}$   
Find:  $\hat{n}$



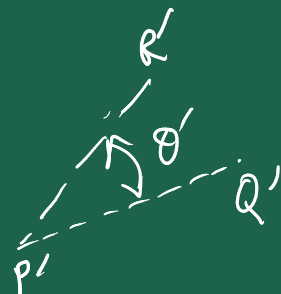
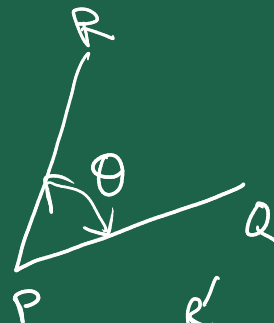
$$d\vec{x} = d\vec{x} + d\vec{x} \cdot \nabla \vec{u}$$

$$\Rightarrow |d\vec{x}| \hat{n} = |d\vec{x}| \hat{N} + |d\vec{x}| \hat{N} \cdot \nabla \vec{u}$$

$$\Rightarrow \frac{|d\vec{x}|}{|d\vec{x}|} \hat{n} = \hat{N} + \hat{N} \cdot \nabla \vec{u}$$

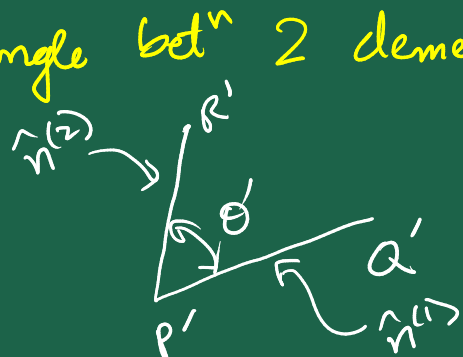
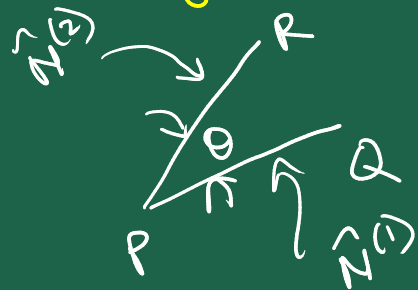
$$\Rightarrow (1 + \epsilon_N) \hat{n} = \hat{N} + \hat{N} \cdot \nabla \vec{u}$$

$$\hat{n} = \frac{1}{1 + \epsilon_N} (\hat{N} + \hat{N} \cdot \nabla \vec{u})$$



$$\theta - \theta'$$

# Change in the angle bet<sup>n</sup> 2 elemental line segments



Given:  $\theta$  between  $\hat{N}^{(1)}$  &  $\hat{N}^{(2)}$

Find:  $\theta'$  between  $\hat{n}^{(1)}$  &  $\hat{n}^{(2)}$

$$\hat{n}^{(1)} \cdot \hat{n}^{(2)} = \left\{ \frac{1}{(1 + \epsilon_N^{(1)})} \left( \hat{N}^{(1)} + \hat{N}^{(1)} \cdot \nabla \vec{u} \right) \right\} \cdot \left\{ \frac{1}{(1 + \epsilon_N^{(2)})} \left( \hat{N}^{(2)} + \hat{N}^{(2)} \cdot \nabla \vec{u} \right) \right\}$$

$$\begin{aligned} \Rightarrow (1 + \epsilon_N^{(1)}) (1 + \epsilon_N^{(2)}) \cos \theta' &= \left[ \hat{N}^{(1)} + \hat{N}^{(1)} \cdot \nabla \vec{u} \right]^T \left[ \hat{N}^{(2)} + \hat{N}^{(2)} \cdot \nabla \vec{u} \right] \\ &= \left\{ \left[ \hat{N}^{(1)} \right]^T + \left( \left[ \nabla \vec{u} \right] \left[ \hat{N}^{(1)} \right] \right)^T \right\} \cdot \left\{ \left[ \hat{N}^{(2)} \right] + \left[ \nabla \vec{u} \right] \left[ \hat{N}^{(2)} \right] \right\} \\ &= \left[ \hat{N}^{(1)} \right]^T \left[ \hat{N}^{(2)} \right] + \left[ \hat{N}^{(1)} \right]^T \left[ \nabla \vec{u} \right] \left[ \hat{N}^{(2)} \right] + \left[ \hat{N}^{(1)} \right]^T \left[ \nabla \vec{u} \right]^T \left[ \hat{N}^{(2)} \right] + \text{h.o.t} \\ \cos \theta' &\approx \cos \theta + \left[ \hat{N}^{(1)} \right]^T \left( \left[ \nabla \vec{u} \right] + \left[ \nabla \vec{u} \right]^T \right) \left[ \hat{N}^{(2)} \right] \end{aligned}$$

$$\cos \theta' \approx \cos \theta + [\hat{N}^{(1)}]^T (\nabla \vec{u}) + [\nabla \vec{u}]^T [\hat{N}^{(2)}]$$

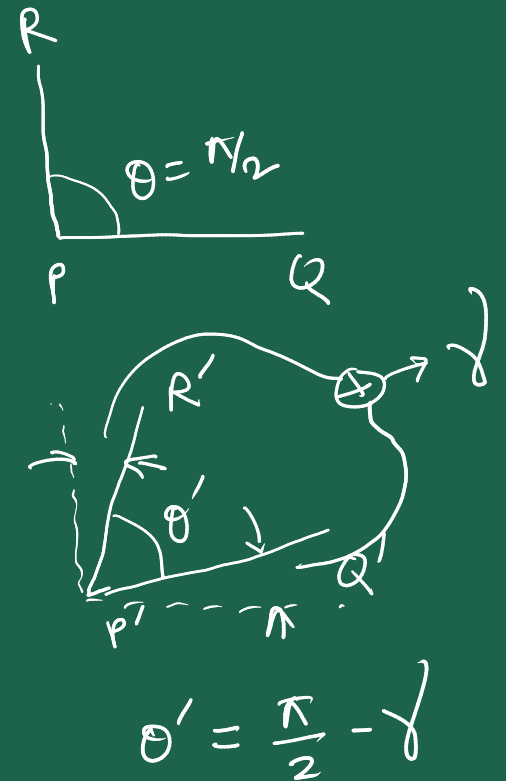
$$\Rightarrow \cos \theta' = \cos \theta + 2 [\hat{N}^{(1)}]^T [\underline{\varepsilon}] [\hat{N}^{(2)}]$$

When  $\theta = \pi/2$

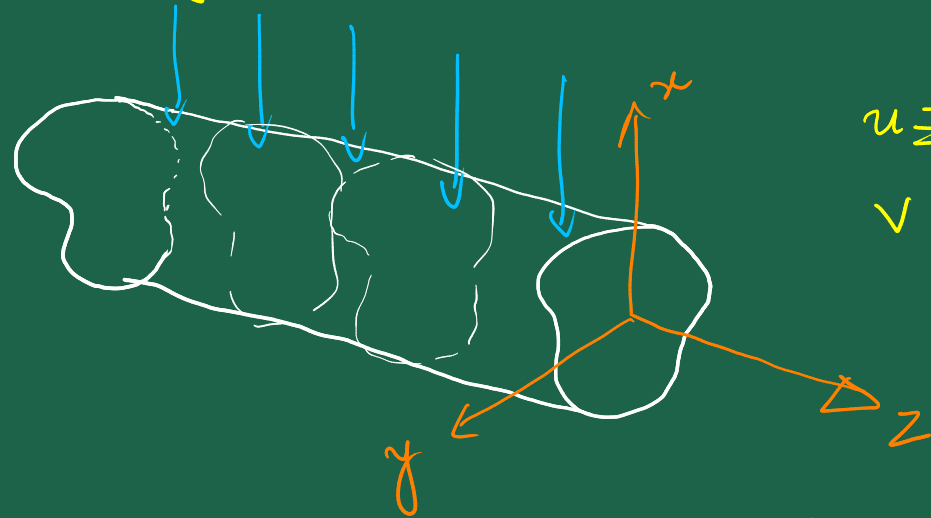
$$\cos \left( \frac{\pi}{2} - \gamma \right) = 0 + 2 [\hat{N}^{(1)}]^T [\underline{\varepsilon}] [\hat{N}^{(2)}]$$

$$\Rightarrow \sin \gamma \approx \gamma = 2 [\hat{N}^{(1)}]^T [\underline{\varepsilon}] [\hat{N}^{(2)}]$$

$$\varepsilon_N = [\hat{N}]^T [\underline{\varepsilon}] [\hat{N}]$$



# Simplifications under the consideration of PLANE STRAIN

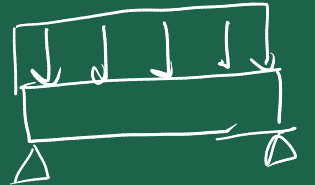
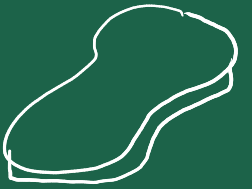


$$u \equiv u(x, y)$$

$$v \equiv v(x, y)$$

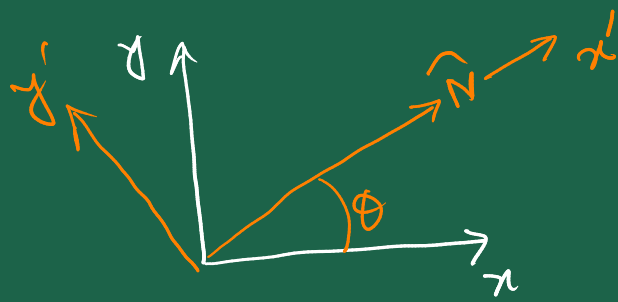
PLANE STRAIN:  $\epsilon_{zz} = 0$ ,  $\epsilon_{zx} = 0$ ,  $\epsilon_{yz} = 0$

$$\begin{bmatrix} \epsilon \\ \tilde{\epsilon} \end{bmatrix} = \begin{bmatrix} \epsilon_{xx} & \epsilon_{xy} & 0 \\ \epsilon_{xy} & \epsilon_{yy} & 0 \\ 0 & 0 & 0 \end{bmatrix}$$



$$\varepsilon_H = [\hat{N}]^T [\varepsilon] [\hat{N}]$$

$$[\hat{N}] = \begin{bmatrix} \cos \theta \\ \sin \theta \end{bmatrix}$$



$$\varepsilon_{x'/x'} = \begin{bmatrix} \cos \theta & \sin \theta \end{bmatrix} \begin{bmatrix} \varepsilon_{xx} & \varepsilon_{xy} \\ \varepsilon_{xy} & \varepsilon_{yy} \end{bmatrix} \begin{bmatrix} \cos \theta \\ \sin \theta \end{bmatrix}$$

$$\frac{d\varepsilon_{x'/x'}}{d\theta}$$

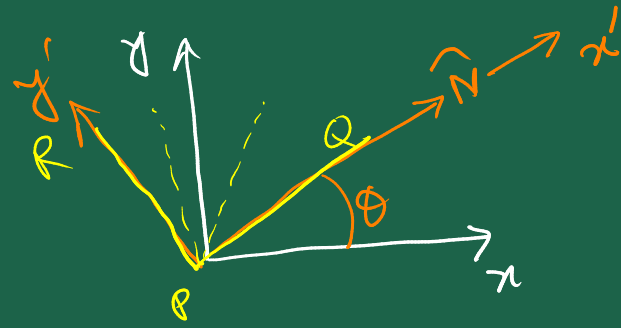
$$\varepsilon_{x'/x'} = \frac{\varepsilon_{xx} + \varepsilon_{yy}}{2} + \frac{\varepsilon_{xx} - \varepsilon_{yy}}{2} \cos 2\theta + \varepsilon_{xy} \sin 2\theta$$

$$\varepsilon_{y'/y'} = \frac{\varepsilon_{xx} + \varepsilon_{yy}}{2} - \frac{\varepsilon_{xx} - \varepsilon_{yy}}{2} \cos 2\theta - \varepsilon_{xy} \sin 2\theta$$

$$\gamma = 2 [\hat{N}^{(1)}] [\tilde{\epsilon}] [\hat{N}^{(2)}]$$

$$[\hat{N}^{(1)}] = \begin{bmatrix} \cos \theta \\ \sin \theta \end{bmatrix}$$

$$[\hat{N}^{(2)}] = \begin{bmatrix} -\sin \theta \\ \cos \theta \end{bmatrix}$$



$$\gamma_{x'y'} = 2 \begin{bmatrix} \cos \theta & \sin \theta \end{bmatrix} \begin{bmatrix} \epsilon_{xx} & \epsilon_{xy} \\ \epsilon_{xy} & \epsilon_{yy} \end{bmatrix} \begin{bmatrix} -\sin \theta \\ \cos \theta \end{bmatrix}$$

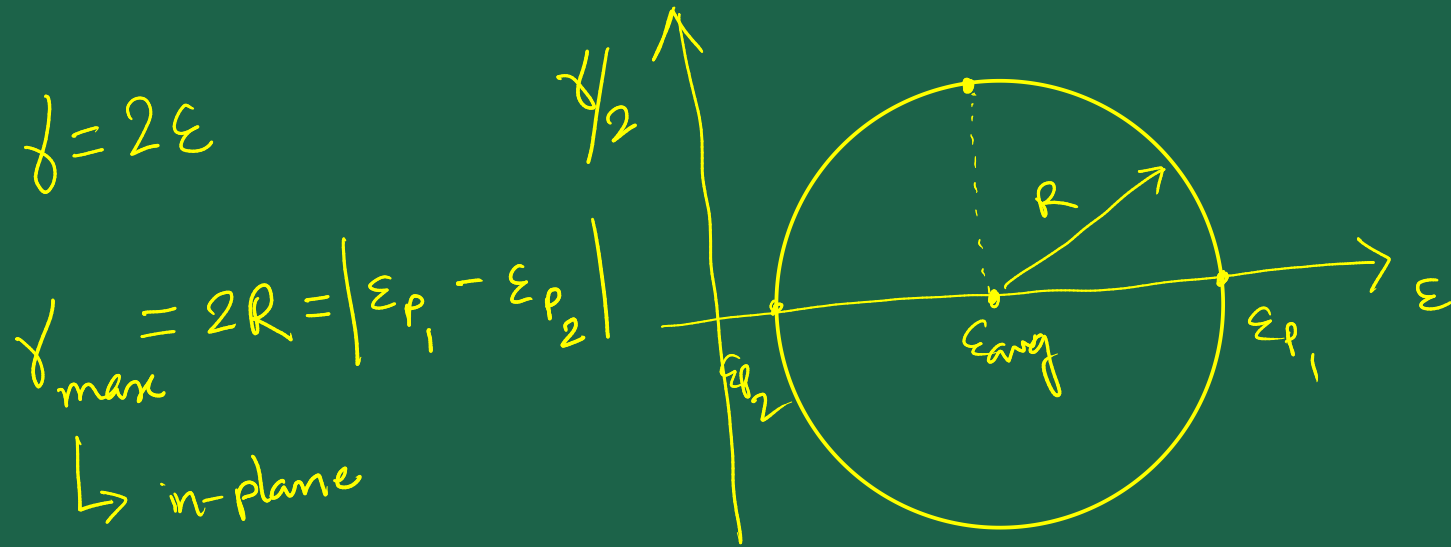
$$= - \left( \frac{\epsilon_{xx} - \epsilon_{yy}}{2} \right) \sin(2\theta) + \epsilon_{xy} \cos(2\theta)$$

$$\left[ \gamma_{xy} = 2 \epsilon_{xy} \right]$$

# Principal strains

$$\epsilon_{p_1, p_2} = \frac{\epsilon_{xx} + \epsilon_{yy}}{2} \pm \sqrt{\left(\frac{\epsilon_{xx} - \epsilon_{yy}}{2}\right)^2 + \epsilon_{xy}^2}$$

← Shamelessly lifted this from principal stress formulae



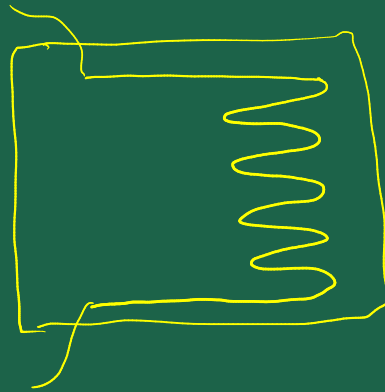
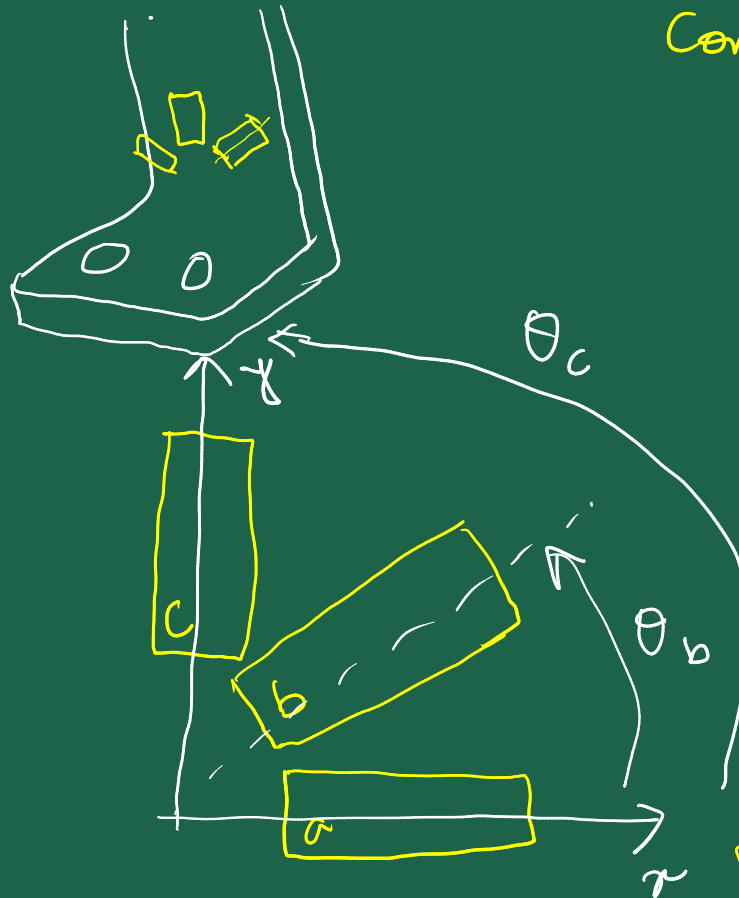
In 3D

$$\gamma_{\max, \text{abs}} = \max \left( |\epsilon_{p_1} - \epsilon_{p_2}|, |\epsilon_{p_2} - \epsilon_{p_3}|, |\epsilon_{p_3} - \epsilon_{p_1}| \right)$$

$$\epsilon_{\text{avg}} = \frac{\epsilon_{xx} + \epsilon_{yy}}{2}, \quad R = \sqrt{\left(\frac{\epsilon_{xx} - \epsilon_{yy}}{2}\right)^2 + \epsilon_{xy}^2}$$

# STRAIN ROSETTE

Combination of strain gages



$$R = \frac{l A}{A}$$

$$\epsilon_a = \epsilon_{xx} \cos^2 \theta_a + \epsilon_{yy} \sin^2 \theta_a + 2\epsilon_{xy} \sin \theta_a \cos \theta_a$$

$$\epsilon_b = \epsilon_{xx} \cos^2 \theta_b + \epsilon_{yy} \sin^2 \theta_b + 2\epsilon_{xy} \sin \theta_b \cos \theta_b$$

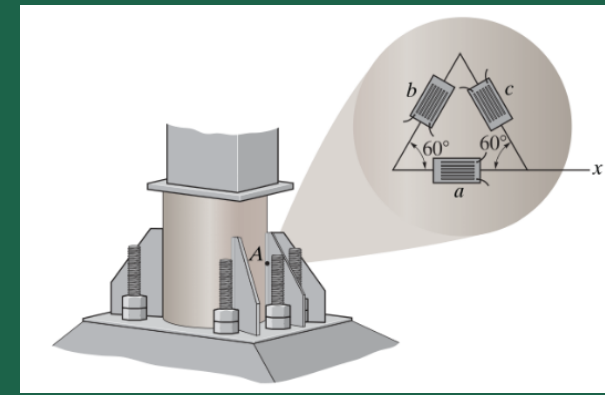
$$\epsilon_c = \epsilon_{xx} \cos^2 \theta_c + \epsilon_{yy} \sin^2 \theta_c + 2\epsilon_{xy} \sin \theta_c \cos \theta_c$$

Linear simultaneous eqns with unknowns  $\epsilon_{xx}$ ,  $\epsilon_{yy}$  &  $\epsilon_{xy}$



4. The strain rosette is attached to point A on the surface of the support. The readings from the strain gauges are:  $\varepsilon_a = 300\mu$ ,  $\varepsilon_b = -150\mu$ , and  $\varepsilon_c = -450\mu$ . Determine (a) the in-plane principal strains, and (b) the maximum in-plane shear strain and (c) the average normal strain associated with the maximum in-plane shear strain. Specify the orientation of each element that has these states of strain with respect to the  $x$ -axis.\*

[ (a)  $336\mu$ ,  $11.7^\circ$ ;  
 $-536\mu$ ,  $101.7^\circ$ ;  
 (b)  $872\mu$ ,  $-33.3^\circ$   
 (c)  $-100\mu$  ]



$$\theta_a = 0^\circ, \theta_b = 60^\circ, \theta_c = 120^\circ$$

$$336 \times 10^{-6}$$

$$\rightarrow \varepsilon_{xx}, \varepsilon_{yy}, \varepsilon_{xy}$$

$$(c) \quad \frac{\varepsilon_{xx} + \varepsilon_{yy}}{2} \quad \text{or} \quad \frac{\varepsilon_{p_1} + \varepsilon_{p_2}}{2}$$

??

We will come back to it!

$$\tan 2\theta_p = \frac{2\varepsilon_{xy}}{\varepsilon_{xx} - \varepsilon_{yy}}$$

$$\theta_{p_1} = 11.705^\circ, \theta_{p_2} = \theta_{p_1} + 90^\circ$$

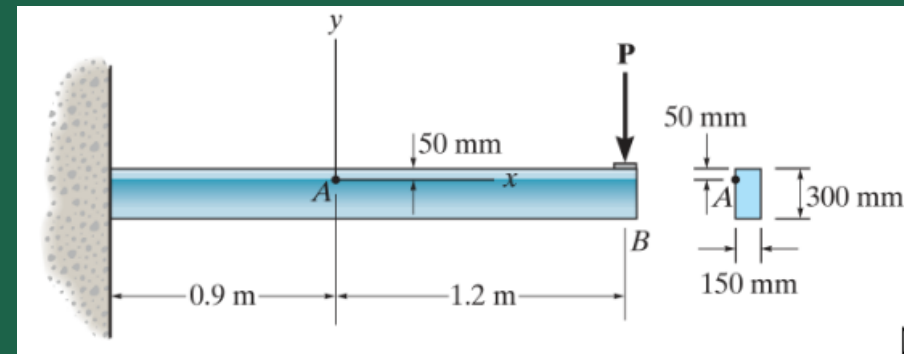
$$\tan 2\theta_s = -\frac{\varepsilon_{xx} - \varepsilon_{yy}}{2\varepsilon_{xy}}$$

$$\theta_s$$



$$\begin{aligned} & \frac{\sigma_{xx} + \sigma_{yy}}{2} \\ &= \frac{\sigma_{x'x'} + \sigma_{y'y'}}{2} \\ &= \frac{\sigma_{p_1} + \sigma_{p_2}}{2} \end{aligned}$$

9. The strain the  $x$ -direction at point A on the structural steel beam ( $E = 203 \text{ GPa}$  and  $G = 76 \text{ GPa}$ ) is measured and found to be  $\varepsilon_{xx} = 100\mu$ . Determine the applied load  $P$ . What is the shear strain  $\gamma_{xy}$  at point A? [ $P = 57 \text{ kN}$ ;  $\gamma_{xy} = -13.91\mu$ ]



$\varepsilon_{xx}$  ✓

$P \rightarrow$  Bending Moment  $\rightarrow$  Flexural stress (or Bending stress)

$$\sigma = \frac{My}{I} \quad M \leftarrow P$$

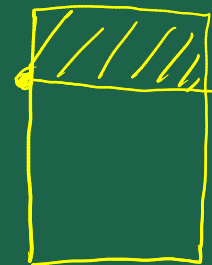
$$\sigma_{xx} = E \varepsilon_{xx}$$

$$\frac{My}{I} = E \varepsilon_{xx}$$

$$\tau_{xy} = \frac{VQ}{It}$$

$$\gamma_{xy} = 2\varepsilon_{xy} = \phi \frac{z_{xy}}{Rb} \quad \checkmark$$

$$\varepsilon_{xx} = \frac{1}{E} \left[ \sigma_{xx} - \nu(\sigma_{yy} + \sigma_{zz}) \right]$$



8. Deduce that in the case of plane strain ( $xy$ -plane) for a body made of a material that follows the generalized Hooke's law, the stress component  $\sigma_{zz}$  itself is a principal stress.

$$\varepsilon_{zz} = 0, \quad \varepsilon_{yz} = 0, \quad \varepsilon_{zx} = 0$$

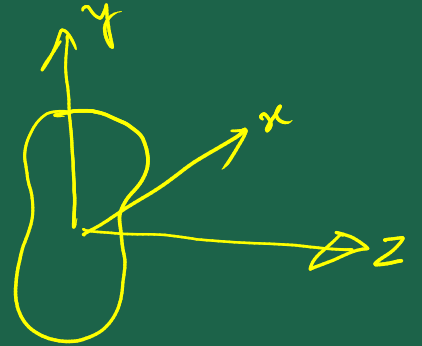
$$\tau_{yz} = 2G \varepsilon_{yz} = 0$$

$$\tau_{zx} = 2G \varepsilon_{zx} = 0$$

$$\sigma_{zz} ?$$

$$\underbrace{\varepsilon_{zz}}_0 = \frac{1}{E} \left[ \sigma_{zz} - \nu (\underbrace{\sigma_{xx}}_{\neq 0} + \underbrace{\sigma_{yy}}_{\neq 0}) \right]$$

$$\Rightarrow \sigma_{zz} \neq 0$$



6. For a material that behaves according to the generalized Hooke's law:

- (a) Considering the case of plane stress ( $xy$ -plane), derive the strain transformation equations from the stress transformation equations.
- (b) How does the strain component  $\varepsilon_{zz}$  transform in part (a)?
- (c) Considering the case of plane strain ( $xy$ -plane), derive the stress transformation equations from the strain transformation equations.
- (d) How does the stress component  $\sigma_{zz}$  transform in part (c)?

(d)  $\sigma_{zz} \neq 0$

observe that  $\sigma_{z'z'}$  will coincide with  $\sigma_{zz}$

(a) Plane stress:  $\sigma_{zz} = 0, \tau_{zx} = 0, \tau_{yz} = 0$

$$\sigma_{x'x'} = \sigma_{xx} \cos^2 \theta + \sigma_{yy} \sin^2 \theta + 2\tau_{xy} \sin \theta \cos \theta$$

$$\varepsilon_{xx} = \frac{1}{E} [\sigma_{xx} - \nu(\sigma_{yy} + \sigma_{zz})]$$

$$\varepsilon_{yy} = \frac{1}{E} [\sigma_{yy} - \nu(\sigma_{xx} + \sigma_{zz})]$$

$$\sigma_{xx} = f_1(\varepsilon_{xx}, \varepsilon_{yy})$$

$$\sigma_{yy} = f_2(\varepsilon_{xx}, \varepsilon_{yy})$$

We'll obtain:  $\varepsilon_{x'a'} = \varepsilon_{xx} \cos^2 \theta + \varepsilon_{yy} \sin^2 \theta + 2\varepsilon_{xy} \sin \theta \cos \theta$

(b)  $\varepsilon_{zz} = \frac{1}{E} [\sigma_{zz} - \nu(\sigma_{xx} + \sigma_{yy})] \neq 0$

Observe that  $\varepsilon_{z'z'}$  is the same as  $\varepsilon_{zz}$ .